CBCS Scheme

		MAY MANAGEMENT
USN		15AE43
		1.0%

Fourth Semester B.E. Degree Examination, Dec.2017/Jan.2018 Aircraft Propulsion

		No. of the second second				
Time:	3 hrs.	Max. Marks: 80				
	Note: Answer FIVE full questions, choosing one full	question from each module.				
1 .	Module-1					
1 a.	1 51	(10 Marks)				
b.	Write difference between four stroke engine and two st	troke engine. (06 Marks)				
	OR OR					
2 a.	Write advantage of gas turbine engine over reciprocation	ng engine (09 Mawka)				
	Define the following:	ng engine. (08 Marks)				
0.	(i) Diffuser efficiency (ii) Compressor efficiency	(iii) Combustion efficiency				
	(iv) Turbine efficiency	(08 Marks)				
		(00 IVIAIRS)				
	Module-2					
3 a.	TYT *: 4 .					
	(i) Variable pitch propeller.					
	(ii) Constant speed propeller.	(06 Marks)				
b.	Explain momentum theory of propeller.	(10 Marks)				
		Σ. Σί				
	OR					
4 a.	1 1					
b.	The difference of the differen					
	adiabatically to 182°C and four times the pressure. Products of combustion enter the turbine					
	at 815°C and leave it at 650°C to enter the nozzle. Calculate the isentropic efficiency of the					
	compressor, power required to drive the compressor, a	and exit speed of gases when flying at				
	800 km/hr. Assume the isentropic efficiency of turbine					
	nozzle efficiency is 90%.	(10 Marks)				
	Module-3	(-2) ,				
5 a.	Derive a relation for minimum area ratio $\left(\frac{A_{max}}{A_{max}}\right)$	in term of external deceleration and				
	A_1	in term of external acceleration and				
	co-efficient of pressure.	(12 Marks)				
b.	and the state of the	(04 Marks)				
	OR					
6 a.	Write short notes on:					
	(i) Nozzle choking.					

(ii) Nozzle throat condition.

(08 Marks)

b. Explain types of thrust reverser. (08 Marks)

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7	a. b.	Module-4 List and explain performance characteristics of centrifugal compressor. Explain rotating stall with sketch.	(10 Marks) (06 Marks)
		OR	
8	a. b.	Define and derive expression for degree of reaction of axial flow compressor. Difference between axial flow compressor and centrifugal compressor.	(10 Marks) (06 Marks)
9	a. b. c.	Module-5 Write advantage and disadvantage of annular combustor. Explain about flame tube cooling. Define combustion intensity.	(06 Marks) (06 Marks) (04 Marks)
10	a. b.	OR Explain turbine blade cooling with sketch. Define: (i) Loading coefficient (ii) Flow coefficient.	(12 Marks) (04 Marks)